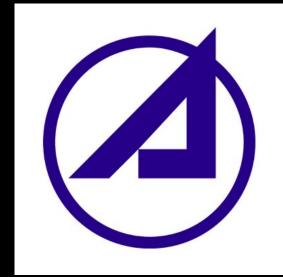
# **Space Wire Bender**



Team Members: Brandon Peffer, Nicolas Adams, Zeke Blanco, Raul Gutierrez, Oscar Rodriguez Faculty Advisor: Dr. Micheal Thorburn Aerospace Corporation Liaisons: Horace Lee, Antonela Pinola Department of Mechanical Engineering College of Engineering, Computer Science, and Technology California State University, Los Angeles



### Background

The COSMIC Capstone Challenge asked students to develop a conceptual design of a payload hosted on board a BCT X-Sat Venus Class Bus. This payload must demonstrate an ISAM (In-space Servicing, Assembly, and Manufacturing) capability using three operations. The team researched problems and noticed risks with large-scale construction in space and the sustainability of current design principles in space. Most satellites are designed to function and then be deorbited once they malfunction or are damaged, are exceedingly expensive to make, and are not intended to be repaired.

# Objective

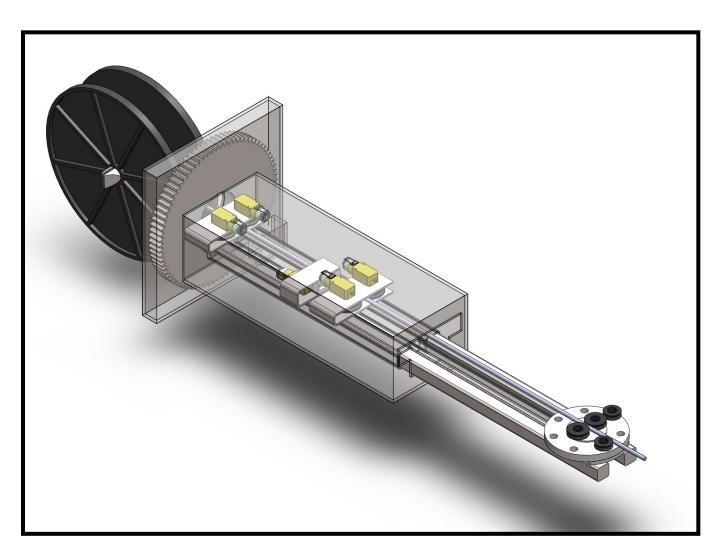
Construction of space structures is expensive and time-consuming. To solve these problems, a device is proposed that bends wire to form structures quickly, accurately, and autonomously in LEO for large-scale space construction. This device must fit within the fairing of a BCT X-Sat Venus Class satellite. It must draw wire from a spool, feed that wire into position accurately and quickly, and complete bends to fabricate truss blocks and other shapes as required.

## **Space Wire Bender Design**

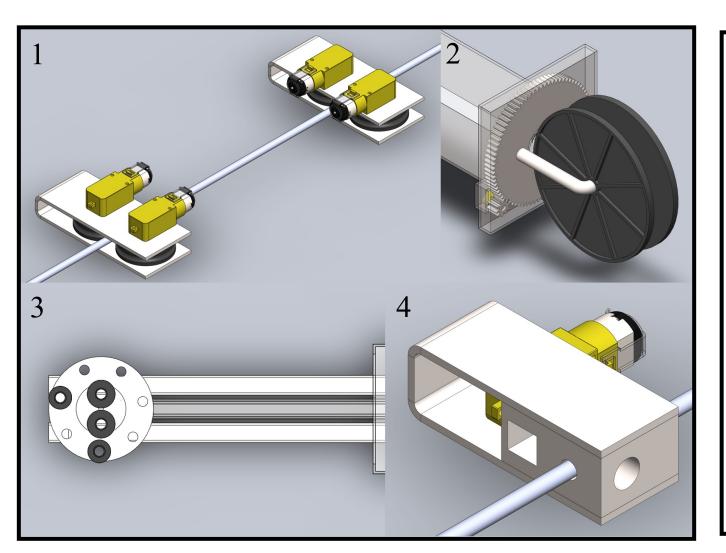
## System & Controls

#### Subsystems

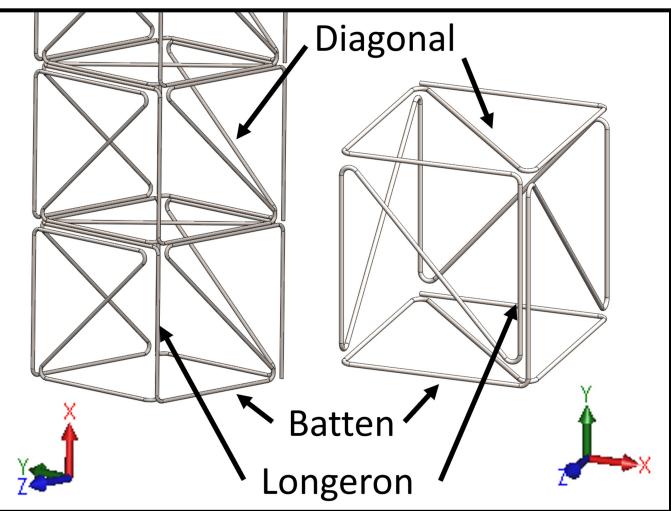
## **Truss Blocks**



- The final design is 27"x11"x11" in compact form, weighs 61 kg and aims to use 220-240W of power
- The technology's conceptual design highlights that it is created for 6 years operational life in LEO (Low Earth Orbit)
- Material selection is based on preexisting and approved satellite materials (E.g., Ti-6Al-4V shell, Kapton insulation, and D2 steel bender heads)
- Low risks mitigated by design, critical failure risks mitigated by redundancy of motors and failsafe operations monitored by sensors
- Overall technological readiness level of 2 highlighting a completed conceptual design for



- 1. Feed
  - Rollers pull the material from the spool into place for other processes along with the bending head deployment allowing for accurate positioning
- 2. Rotator
  - Large gear located on the back of the device holds the spool of material, this allows the spool to be rotated 360 degrees
- 3. Bender
  - A rotary draw bender bends the material to the desired angle in both clockwise and counterclockwise directions
- 4. Cutter
  - An enclosed cutter cuts the wire so the



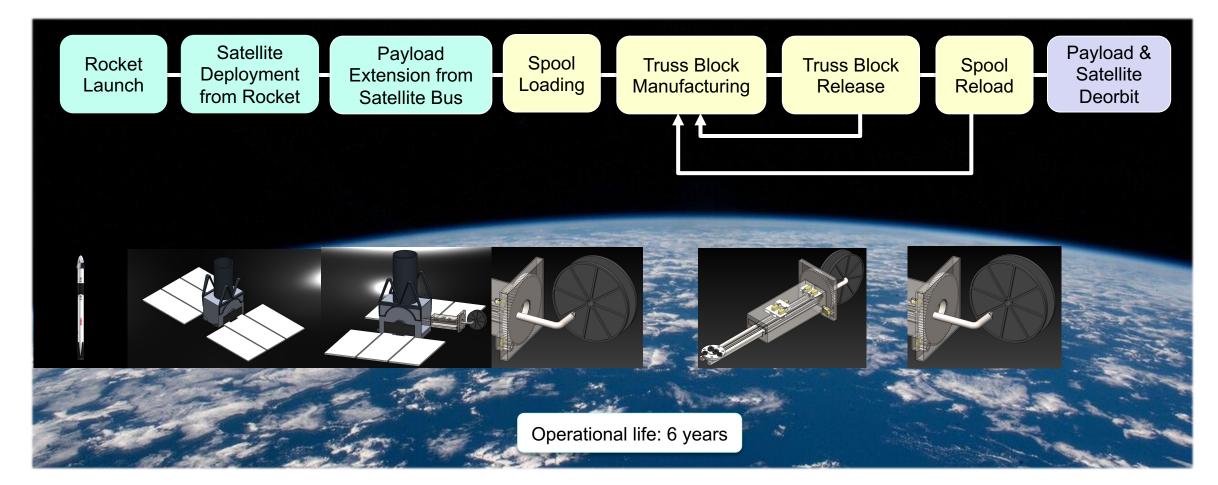
- 316 SS wire, AWG 4 (0.204") dia.
- SmallSat structures, supports & fasteners
- Modular Truss Blocks
  - Blocks (3D): cube, pyramid
  - Panels (2D): rectangle, triangle
  - Complex shapes
- Supports
  - Brackets, connectors
- Fasteners
  - Wire ties
  - Wire snap fits

initial stepping-stone of ISAM ecosystem

final truss can be removed from the device to be used in other processes. Contains and stores all generated debris within

- Application & Impact
  - SmallSat booms, frame panels
  - SmallSat factory
  - Scaled up to larger satellites

### **Projected Operation Lifecycle**



#### **The Team**

**Zeke Blanco** 



**Brandon Peffer** 



Nicolas Adams



**Raul Gutierrez** 



**Oscar Rodriguez**